Program Update 10-Year Lunar Architecture (LunA-10) Capability Study

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LunA-10 consortium of industry, arranged by (initial) services

These groupings change across the program based on proposed technical areas/products



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Push from individual self-service to <u>commercial multi-service</u>





For a given service or unit: what are the inputs/outputs/limitations?

What DARPA-hard technical challenges must be surmounted to create a sustainable lunar economy by 2035?

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A fast and furious summary of over 25,000 person-hours of labor



Three complementary, multi-service commercial systems:

- 1. Lander node as payload host and Infrastructure Platform
- 2. Laser-enabled Wireless power framework
- 3. ISRU for Construction, Mining and Energy



Features	Capability
Solar Array	> 10 kWe
Mast	20 m mast on ~10 m lander (total 30 m above surface)
3GPP Telecom Service	25 Mbps bps up to > 10 km range, max range ~100 km
Regen Fuel Cell Augmentation Kit	1.5 MWh, 7.8 kW _e over 192 hrs
Laser Power Transmitter	~1 kW _e delivered to 10+ km,
Silicon Extraction Experiment	Demonstrate production of silicon from regolith
Heat Rejection Augmentation Kit	Added Radiator area for payload power









Scalable, commercially commoditized unit for:

Hybrid PNT

Solution

- Surface area networking (SAN),
- Space traffic management (STM),
- Direct-to-Earth comms,

Two-Way Ranging Solutions

- Position Navigation and Timing (PNT),
- Add-ons for night survivability





Fibertek: Lunar Infrastructure Optical Node (LION)

Scalable, modularized unit for surface power and connectivity to mobile users:

- Power beaming across the lunar surface
- High-speed surface optical back-bone
- Direct-to-earth communications link
- Lunar-to-orbit communications link







DARPA

Firefly Space: a cislunar aggregation hub for propellant and spacecraft





Robotic labor charged in \$/hour will:

- Enable the construction of lunar infrastructure including ISRU, power equipment, and transportation networks
- Provide maintenance as components fail or reach end-of-life
- Controllable from earth with no astronaut necessary







Scalable process for Solid Oxide Electrolysis to extract oxygen from regolith

Products include:

- Gaseous O2
- Liquid O2 (LOX) propellant
- De-oxygenated regolith for further metal refinement & processing



O₂ Liquid

Regolith Inlet

1 Part

ELID

RG-1

.....

02

Gas

Hopper

Grapple fixtures for

robotic maintenance

Reactor

LUNARSABER is a utility tower for services including navigation, sensing, and power distribution

- Lightweight, deployable tower capable of hosting elevated payloads
- Increased line-of-sight for comms, power beaming, and PNT while simultaneously collecting solar power

2 -10 -1 -0.2 -0 -0.5 -1 -1.5 -2

19.37%

% Viewable of Local Area (15 km radius)

-25 -2 -18 -1 -05 -0 -05 -1 -15 -2

25 2 15 1 65 0 65

Laser VMX to pave roads and landing pads

- Resource-efficient inputs of raw regolith requires minimal materials launched to surface
- Current production time estimate is ~1 month for a 10m diameter landing pad with 10 kW surface power supplied

ICON's Project Olympus for NASA and commercial lunar projects

6.5 km

17 km

Multi-purpose railway for lunar mobility

Concept designs of rail equipment to include:

- Construction materials and track foundation design
- Railcars for cargo transportation
- Vehicles for construction and maintenance operations

Cross-Mass Mobility: Trip Momentum Cost Efficiency

Terrestrial Transportation

Rover Transportation

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Phase 2

Landing

Lunar Rail Transportation

Constellation of cislunar orbiters to provide:

- Communications
- PNT
- RF Survey
- SAR/MTI
- Microwave space-based solar power beaming

Oxygen ISRU plant with three main functions:

- 1. Oxygen Extraction from Regolith
- 2. Fuel Cell Energy Storage for STN
- 3. Chemical Conversion for Recycling and Storage

Artist concept of a carbothermal oxygen production plant

Starship as both an earth-to-moon launch vehicle and a lunar surface hub

- Super-heavy, reusable transportation vehicle with integrated comms, power, and nighttime battery services
- 3rd party payloads deployable from height of Starship
- Surface hub can provide propellant and fluid transfer services

Designed for reuse and re-launch from lunar surface

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Emerging topics that require further technical development

Man-made equipment on the Moon may grow by an order of magnitude in next 10 years

The definition of "in-situ" resources also includes defunct manmade equipment How do we use that to reduce mass required from Earth?

Mobile lunar systems demand wireless power Tx/Rx: technically feasible

LION Micro

0.74 kW regulated power

1.9 kW radiated heat

Mass: 223 kg, including

LION Mini

3.0 kW regulated power

Mass: 285 kg, including

7.8 kW radiated heat

LION

5.9 kW regulated power

15.5 kW radiated heat

Mass: 360 kg. including

LION Multi

Individual beam directors

SWaP: Scalable, up to full

per laser

Power scalable

Regulated Power to User (kW)

LION Nano

0.35 kW regulated power

0.97 kW radiated heat

Mass: <80 kg. no tower

DARPA

Wireless power (\$/kWh) a viable near-term commercial solution

Inchworm robots coupled with mobile rovers will provide tele-operated robotic labor

Charged based on hourly rate Customizable to a user's construction needs Ideal for building, maintenance, and component replacement Modular and scalable to most lunar construction applications

Foundational Networked towers for Multi-service consolidation

Services on Masthead

- Centralized service node with expanded line of sight across rugged lunar terrain.
- Can service or be integrated with almost any current or future lunar service: power, storage, data, comms, PNT, light, thermal, and more.
- "In one launch" setup paradigm.

Services at Base

DARPAThermal as a Service:Provide heat rejection/generation and change future lunar design paradigms

Thermal management is a need for all users on lunar surface

- Several key ISRU processes run at high temps (>1000C)
- Lunar night challenges to maintain survivable temps (14+ earth days)
- High % of thermal mass required on-launch

A thermal utility service is needed to accommodate the Moon's extreme thermal environment

Figure: Example of a simulation-set run in CFD, 14 Earth days, 800 K Input

DARPA Lunar Rail Network: Provide crossmass capability for development of regional hubs

Cargo transportation to enable expansion of a lunar base

Infrastructure design based on material properties of regolith and anticipated user requirements (LunA-10 partners)

Current solution for crossmass: Rovers

- Limited heavy material-handling capabilities
- Limited infrastructure for storage of volatiles
- Limited travel distances with onboard power

One example set of teaming arrangements (changed every 2 weeks)

Team 1: How would *GITAI* assemble and maintain a *Sierra Space* and *Helios* oxygen plant with robotic labor?

Team 2: Is there demand from a *SpaceX* spacecraft to join a *Firefly* orbital aggregation hub?

Team 3: How would a *Honeybee* solar power tower be designed and integrated with *Fibertek* power beaming and *Crescent* comms systems at the tower's height?

DARPA

Team 4: Do *CisLunar* 's metal products and *ICON* 's sintered regolith have sufficient material properties to construct a *Northrop Grumman* lunar rail network?

Not a fully inclusive list (not even close)

Please attend LunA-10 breakthrough sessions to learn more

www.darpa.mil

Appendix – White Space Charts

CisLunar Industries

CisLunar Industries

Question 4 "White space chart of financial aspect"

In order to keep providing RaaS, The formula "**Cost < Revenue**" should hold. This formula can transform to the following formula.

D + R + T * M < N * L * O * P

In order to keep providing RaaS in lunar commercial market, the price should fit to customer demand price. If the price doesn't meet the customer demand price, we can take two types of strategy as shown on the next page.

Pad Production vs Power on Surface and Time

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Northrop Grumman

Sierra Space

Assumptions: Second Y axis is Rocket Dry Mass / Delta V-Year achievable assuming Mass ratio of 0.5248 (Same as SLS) ISP of 465